# Convective wall plume in power-law fluids

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Abstract-This paper considers the steady-state free convection flow arising from a line thermal source positioned at the leading edge of a vertical adiabatic surface embedded in polymeric fluids. The wall plume depends on two parameters: the index of the power-law fluid  $(n)$  and the generalized Prandtl number  $(Pr)$ . Precise conditions of finding similarity solutions for this problem are derived. A family of numerical solutions for *n* ranging from 0.2 to 2.0 and for  $Pr = 10$  and 100 is reported.

# 1. INTRODUCTION

SIMILARITY solutions for the free convection flow of a Newtonian fluid arising from a steady line thermal source embedded at the leading edge of a vertical adiabatic surface date back to an early paper by Zimin and Lyakhov [I] in 1970.

Later, Jaluria and Gebhart [2] presented accurate numerical solutions of this problem for a Prandtl number range of 0.01-100. Afzal [3] and recently Ingham and Pop [4] have derived higher-order solutions for convective wall plumes for moderately large values of the Grashof number of the method of matched asymptotic expansions. A search of the literature reveals that papers [5-81 are also all devoted to the problem of convective wall plumes in a Newtonian fluid.

The aim of this paper is to investigate the free convection flow of a power-law fluid arising from a line thermal source positioned at the leading edge of a vertical adiabatic surface. A systematic analysis for deriving a possible similarity formulation for this flow problem is presented. In Section 2, the similarity transformation of the boundary layer equations is introduced and numerical solutions are presented for various values of the power-law index,  $n$ , and the generalized Prandtl number,  $Pr$ . The concluding section draws attention to the principal results of this paper.<br>Paper paper.<br>It is worth mentioning at this point that the trans-

port phenomenon in power-law fluid flow has been port phenomenon in power-law huid how has been the subject of many recent investigations due to the frequent use of this type of fluid in modern industry. Several review articles and books may be consulted several feview articles and books may be consulted<br>Containing in the containing for the theory

authors' knowledge no investigation of the problem considered in the present paper has been reported previously in the literature.

### 2. ANALYSIS

Consider the problem of steady, laminar, free convection from a line source of heat positioned at the leading edge of a vertical adiabatic surface immersed in an unbounded power-law fluid with the following transport properties [15, 16]

$$
\tau_{ii} = -P\delta_{ii} + K[0.5J_2]^{(n-1)/2}e_{ii}
$$
 (1)

$$
q = -k[0.5J_2]^{s/2} \text{ grad } T \qquad (2)
$$

where  $\tau_{ii}$  and  $e_{ii}$  are the tensors of stress and strainrate,  $\delta_{ij}$  is unit tensor,  $J_2$  is the second invariant of the strain-rate tensor,  $n$  and  $s$  are superscripts identifying non-Newtonian behaviour in the flow and heat transfer. The strict Boussinesq approximation is assumed, i.e. the variation of fluid density with temperature is accounted for only in the buoyancy term of the vertical momentum equation ; all other fluid properties are assumed to be constant; and viscous dissipation is neglected. The plume is assumed to be laminar and the boundary layer approximation is assumed to hold. We chose  $(x^*, y^*)$  as coordinates with the  $x^*$ -axis measured along the wall in an upward direction and the  $y^*$ -axis is normal to it. The temperature T takes the value  $T_{\infty}$  in the ambient fluid. Under these conditions the basic equations in non-dimensional form (for details see Table 1) can be written as

$$
\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0 \tag{3}
$$

# NOMENCLATURE

 $\alpha$ 

- $C_f$  skin friction coefficient  $U$  reference velocity<br>  $C_n$  specific heat at constant pressure  $x^*, y^*$  Cartesian coordinates.
- $C_p$  specific heat at constant pressure
- $e_{ij}$  strain-rate tensor reduced stream function Greek symbols
- 
- $g$  acceleration due to gravity Gr generalized Grashof number  $\beta$
- $h$  reduced temperature function  $\delta_{ij}$  unit tensor
- $J_2$  second invariant of the strain-rate tensor  $\theta$  dimensionless temperature
- $k$  thermal conductivity  $n$  similarity variable
- $K$  consistency index  $\rho$  density
- $L$  reference length  $\tau$  shear stress
- *n* flow behaviour index  $\psi$  stream function.
- P pressure
- Pr generalized Prandtl number Subscripts
- $q, I$  dimensional and non-dimensional heat  $w$  wall condition input by the thermal source  $\infty$  ambient condition.
- s heat transfer behaviour index
- T temperature Superscripts
- $T_r$  reference temperature differentiation with respect to  $\eta$
- $u^*, v^*$  velocity components \* dimensional variables.

$$
u\frac{\partial u}{\partial x} + v\frac{\partial u}{\partial y} = \frac{\partial}{\partial y} \left( \left| \frac{\partial u}{\partial y} \right|^{n-1} \frac{\partial u}{\partial y} \right) + \theta \tag{4}
$$

$$
u\frac{\partial \theta}{\partial x} + v\frac{\partial \theta}{\partial y} = \frac{1}{Pr}\frac{\partial}{\partial y}\left(\left|\frac{\partial u}{\partial y}\right|^s \frac{\partial \theta}{\partial y}\right).
$$
 (5)

The associated boundary conditions are

$$
y = 0: u = v = 0 \tag{6a}
$$

$$
\theta = (T_{\rm w} - T_{\infty})Gr^b/T_{\rm r} \quad \text{or} \quad \frac{\partial \theta}{\partial y} = 0 \tag{6b}
$$

$$
y \to \infty : u = 0, \quad \theta = 0 \tag{6c}
$$

together with a condition which expresses the fact that there is uniform heat flux from the line source

$$
\int_0^\infty u\theta \, \mathrm{d}y = I \tag{7}
$$

where

$$
I = q(\rho C_p L T_r)^{-1} (\rho L^n G r^{n-1}/K)^{1/(n-2)}.
$$
 (8)

The generalized Grashof number is

$$
Gr = g\beta T_r L^{(2+n)/(2-n)} / (\rho/K)^{2/(n-2)}
$$

and the generalized Prandtl number is

Table 1. Dimensionless variables

$\boldsymbol{x}$		и	$\boldsymbol{\eta}$				
$\frac{x^*}{L}$	$\frac{y^*}{I}$ Gr <sup>a</sup>	$\frac{u^*}{U}$	$\frac{v^*}{U}$ Gr <sup>a</sup>	$\frac{T-T_{\infty}}{T}Gr^b$			
$a = n/(4n+1), b = (6n-5n-2)/[(4n+1)(n-2)]$							

$$
Pr = L^{1+s}U^{1-s} Gr^{-n(2+s)/(1+4n)}/\alpha
$$

where

(5) 
$$
U = (\rho L^{n} Gr^{-n(n+1)/(4n+1)}/K)^{1/(n-2)}
$$

is the characteristic velocity for this free convection situation.

The solution of the coupled non-linear partial differential equations  $(3)$ – $(5)$  is facilitated by a number of transformations. The first step is to introduce a stream function  $\psi$  such that

$$
u = \frac{\partial \psi}{\partial y}, \quad v = -\frac{\partial \psi}{\partial x}.
$$
 (9)

Then, the pseudo-similarity variables are defined as

$$
\psi = x^{(2n+1)/(4n+1)} f(x, \eta), \quad \theta = x^{-(2n+1)/(4n+1)} h(x, \eta)
$$
\n(10a)

and

$$
\eta = x^{-(n+1)/(4n+1)}y.
$$
 (10b)

The temperature of the wall is assumed to depend upon  $x$  in the following manner:

$$
T_{\rm w}(x) = T_{\infty} + Gr^{-b} T_{\rm r} x^{-(2n+1)/(4n+1)}.
$$
 (11)

Insertion of equation (10) into equations  $(3)$ – $(5)$  leads to

$$
(|f''|^{n-1}f'')' + \frac{2n+1}{4n+1}ff'' - \frac{n}{4n+1}f'^2
$$

$$
+ h = x\left(f'\frac{\partial f'}{\partial x} - f''\frac{\partial f}{\partial x}\right) \quad (12)
$$

r

thermal diffusivity

thermal expansion coefficient

$$
x^{(n-1-s)/(4n+1)} \frac{1}{Pr} (|f''|^s h')' + \frac{2n+1}{4n+1} (fh)'
$$
  
=  $x \left( f' \frac{\partial h}{\partial x} - h' \frac{\partial f}{\partial x} \right)$  (13)

subjected to the boundary conditions

$$
f(x, 0) = f'(x, 0) = 0
$$

$$
h(x, 0) = 1 \quad \text{or} \quad h'(x, 0) = 0 \tag{14a}
$$

$$
f'(x, \infty) = h(x, \infty) = 0 \tag{14b}
$$

where primes denote partial differentiation with respect to  $\eta$ .

It is apparent that these equations will permit similarity solutions if the exponent of x in equation (13) vanishes, i.e.

$$
s = n - 1. \tag{15}
$$

Under this restrictional condition, equations (12) and (13) become

$$
(|f''|^{n-1}f'')' + \frac{2n+1}{4n+1}ff'' - \frac{n}{4n+1}f'^2 + h = 0
$$
\n(16)

$$
\frac{1}{Pr}(|f''|^{n-1}h')' + \frac{2n+1}{4n+1}(fh)' = 0
$$
 (17)

with the boundary conditions of

$$
f(0) = f'(0) = 0, \quad h(0) = 1, \quad h'(0) = 0 \tag{18a}
$$

$$
f'(\infty) = h(\infty) = 0. \tag{18b}
$$

The heat flux condition (7) may now be written as

$$
I = \int_0^\infty f'h \, \mathrm{d}\eta = I(n, Pr). \tag{19}
$$

We notice to this end that for  $n = 1$ , equations (16) and (17) reduce to those of refs. [3, 41 which describe the classical problem of convective wall plume in a Newtonian fluid.

It is customary to present the flow characteristics by means of the skin friction coefficient

$$
C_{\rm f} = 2 \frac{\tau_{\rm w}}{\rho U^2}.
$$
 (20)

Making use of equations (I), (9) and (IO) the skin friction coefficient (20) becomes

$$
C_f \; Gr^{n/(4n+1)} \; x^{n/(4n+1)} = 2 |f''(0)|^n. \tag{21}
$$

In conclusion, the problem to be solved is that pre- $\frac{1}{2}$  by equations (I  $\frac{1}{2}$ ).  $\frac{1}{2}$ 

# 3. RESULTS AND DISCUSSION

 $E_{\rm eff}$  and (large to the subject to the  $\mathbf{b}$  between  $\mathbf{b}$  and  $\mathbf{b}$ , which are subject to the boundary conditions (18), have been integrated numerically by using the Runge-Kutta-Gill method for *n* ranging from 0.2 to 2.0 and for  $Pr = 10$  and 100, respectively. As mentioned in Section 2, in this model

Table 2. Numerical values of computed parameters for various *n* and  $Pr = 10$ 

	n f''(0)	$\sim I$ .	$\int_{\max}$	$f(\infty)$ $\eta(f'_{\max})$		$\eta_{0.50}$
		0.2 3.21309 14.67177 0.70945 3.46476			1.8000	3.25674
04	1.56230		9.05100 0.56938 2.48399		1.6500	2.15474
	0.6 1.08804		5.18968 0.45349 1.52345		1.3000	1.49193
0.8	0.95063		3.96055 0.43117 1.32087		1.2000	1.24216
	1.0 0.86123		3.00209 0.39276 0.99482		1.0000	1.04663
	1.2 0.90450		2.87300 0.49054 1.62756		1.4000	0.92290
	1.5 0.82903		2.12471 0.40676 0.70861		0.9000	0.82179
	2.0 0.76044		1.27685 0.30440 0.32992		0.6000	0.64404

*n* is the property of a fluid with  $n = 1$  for a Newtonian fluid. Non-Newtonian fluids with  $n < 1$  are called pseudo-plastic (most macromolecular fluids are of this kind with  $0.2 < n < 0.6$ , see Bird *et al.* [17]) and those with  $n > 1$  dilatant.

The results for various transport parameters, which are important for representing the heat transfer correlations, see Gebhart et al. [18], are given in Tables 2 and 3 for the flow behaviour index ranging from 0.2 to 2.0 and Prandtl numbers of IO and 100, respectively. In order to assess the accuracy of our numerical results, the present results were compared with those of ref. [4] for  $n = 1$  (Newtonian fluids). Thus, the values for  $C_1 Gr^{1/5} x^{1/5}$  from equation (21) when  $n = 1$ are 2.62012 for  $Pr = 0.72$  and 1.85964 for  $Pr = 6.7$ while the corresponding values from ref. [4] are 2.6201 and 1.8596, respectively. This shows that the agreement is excellent. Also, comparison with the results of Liburdy and Faeth [19] for  $n = 1$  with  $Pr = 10$  and 100 is found to be very good.

It is noted that from the present results that the friction factor decreases with increasing values of  $n$ and Pr. This fact is also verified from the results presented in Fig. 1. The integral  $I$  defined in equation (19) determines the velocity level and the surface temperature. From Table 3 we observe that I decreases with increasing values of  $Pr$  and  $n$ . Note that having determined the values of I, the reference temperature  $T<sub>r</sub>$  can be obtained easily through equation (8).

Figures 2 and 3 display results for the upward velocity profiles in the wall plume. It is observed that the maximum velocity decreases with increasing values of the flow behaviour index,  $n$ . The location of the maximum velocity moves closer to the wall as  $n$ 

 $T$ able 3. Numerical values of computed parameters for computed parameters for computed parameters for computed parameters for  $\lambda$ cal values of computed

	n f''(0)	$\mathcal{L}$	$f'_{\text{max}}$ $f(\infty)$ $n(f'_{\text{max}})$		$\eta_{0.50}$
0.2 <sub>1</sub>	1.23283 3.33278 0.22884 0.90717			1.5000	2.19335
0.4		0.73858 2.33788 0.20504 0.73486		1.3000	0.87350
0.6		0.50828 1.04260 0.13315 0.19407		0.8000	1.02645
08		0.41895 0.53576 0.09548 0.08289		0.5400	0.76715
10		0.48062 0.52386 0.12145 0.12585		0.6000	0.59084
12		0.54364 0.48404 0.16156 0.19320		0.8000	0.49105
15		0.53094 0.31474 0.13440 0.12537		0.5000	0.41985
20		0.44790 0.10934 0.06097 0.01668		0.2100	0.25987



FIG. 1. Friction factor vs flow behaviour index, n, for  $Pr = 10$ and 100.



increases. The boundary region thickness decreases as n increases. As the Prandtl number increases, the thinning effect on the thermal layer influences the boundary region. Also, it is remarkable that the curve for  $n = 1$  (Newtonian fluids) appears to intersect more curves for  $Pr = 100$  than for  $Pr = 10$ . The reason for this seems to be the dependence of the Prandtl number on the flow behaviour index,  $n$ , reference velocity,  $U$ , and the reference length, L, of the vertical surface. The results from Figs. 4 and 5 describe the temperature distribution in the wall plume. It is observed that as Pr or n increases, the thermal layer becomes thinner. The important considerations in this flow are the velocity level, the surface temperature and the extent of

FIG. 3. Upward velocity profile vs similarity variable,  $\eta$ , for  $Pr = 100$ .

the boundary region. As the flow proceeds downstream from a heated element located on an unheated vertical surface, it influences the cooling characteristics of any other elements it may encounter. An element downstream is immersed in a flowing heated fluid, whose temperature and velocity are determined by the distance between the two elements and the heat flux input *I*. Tables 2 and 3 show the necessary values of  $f''(0)$ , I,  $f'_{\text{max}}$ ,  $f(\infty)$  and  $\eta(f'_{\text{max}})$ , over the Pr and n ranges considered, to allow evaluation of the temperature and velocity fields at a downstream element.





FIG. 2. Upward velocity profile vs similarity variable,  $\eta$ , for



profile vs similarity variable,  $\eta$ , for Fig. 4. Temperature distribution in the wall plume vs simi-<br> $Pr = 10$ .<br>larity variable,  $\eta$ , for  $Pr = 10$ .



FIG. 5. Temperature distribution in the wall plume vs similarity variable,  $\eta$ , for  $Pr = 100$ .

line source embedded at the leading edge of a vertical plate in a power-law fluid is an interesting practical problem. It is hoped that the present work will elicit some experiments for quick and yet accurate estimations of convective heat transfer rates.

## 4. CONCLUDING REMARKS

In this paper, we have analysed the laminar natural convection flow generated by a line thermal source imbedded in an adiabatic vertical surface. The flow configuration is of much interest since the governing boundary layer equations admit similarity solutions which are more revealing than the direct numerical integration of the partial differential equations. In addition, this problem is of considerable importance in engineering applications, such as the positioning of components dissipating energy on vertical circuit boards, and the results concerning the boundary layer flow characteristics are reported here. The numerical results presented in this paper allow evaluation of the velocity and temperature fields in the generated flow. The flow behaviour index was varied from 0.2 to 2.0 whereas the Prandtl number was taken as 10 and 100.

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